



Class: 2<sup>nd</sup>

Subject: Strength of Materials

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**Strength of Materials**

**Second Stage**

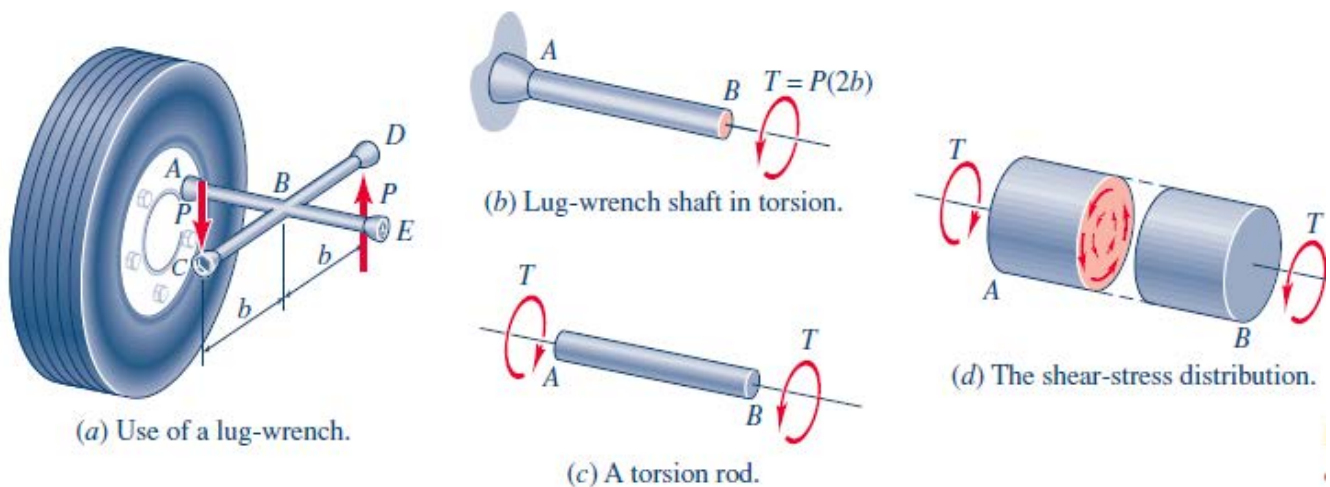
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# Torsion

## Introduction

- ▶ In this chapter we will concentrate on the behavior of slender members subjected to torsional loading that is, loading by couples that produce twisting of the member about its axis.
- ▶ Figure (4.1) shows a common example of torsional loading and indicates the shear stresses and the stress resultant associated with torsion.
- ▶ In Figure (4.1), shaft AB is say a *torsion member*.
- ▶ The shaft AB is subjected to equal and opposite torques of magnitude  $T$  that twist one end relative to the other.



**Figure (4.1): An example of torsion**

- ▶ Several names are applied to torsion members, depending on the application:
  - *shaft*,
  - *torque tube*,
  - *torsion rod*,
  - *Torsion bar*, or
  - *Simply torsion member*.

❑ **Analogy Between Axial Deformation and Torsion**

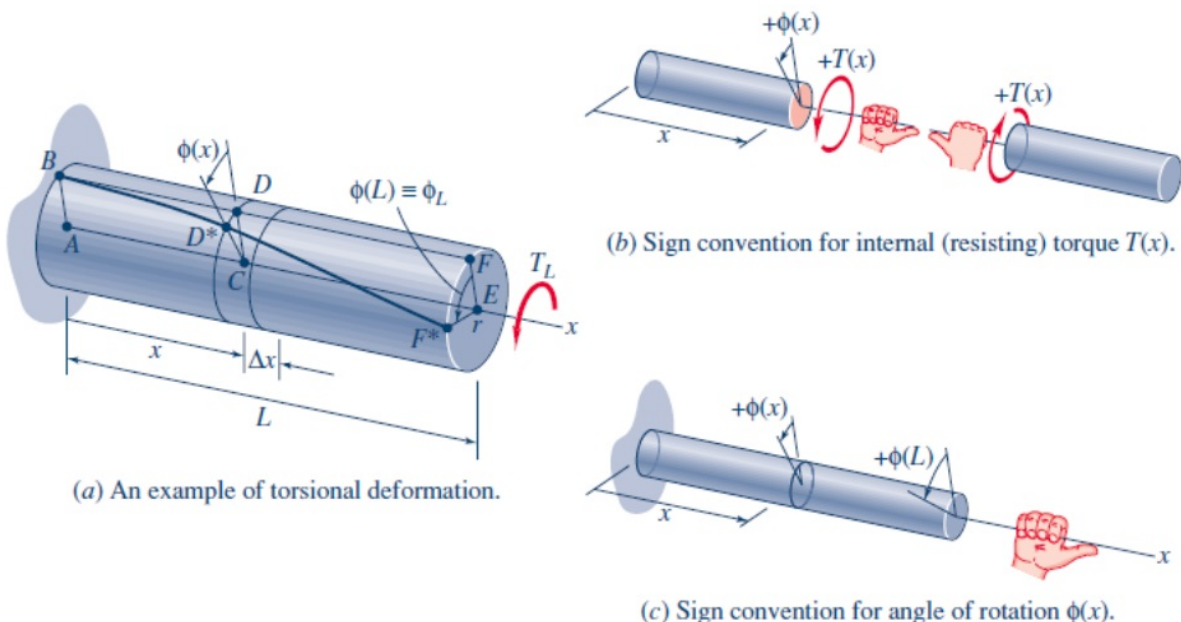
- ▶ There is a direct analogy between axial deformation and torsion, as indicated by the entries in Table 4.1.

| TABLE 4.1 Analogy Between Axial Deformation and Torsion |                           |
|---|---------------------------|
| Axial Deformation                                       | Torsion                   |
| Axial Force ( $F$ )                                     | Torque ( $T$ )            |
| Elongation ( $e$ )                                      | Twist angle ( $\phi$ )    |
| Normal stress ( $\sigma$ )                              | Shear stress ( $\tau$ )   |
| Extensional strain ( $\epsilon$ )                       | Shear strain ( $\gamma$ ) |
| Modulus of elasticity ( $E$ )                           | Shear modulus ( $G$ )     |

**Sign convention**

A **sign convention for torsion** is defined as follows:

- ▶ The longitudinal axis of the bar is labeled the  $x$  axis, with one end of the member being taken as the origin.
- ▶ A positive torque,  $T(x)$ , is a moment that acts on the cross section at  $x$  in a right-hand-rule sense about the outer normal to the cross section. On a cross sectional cut at  $x$  there will be equal and opposite torques  $T(x)$ , as indicated in Figure (4.2b).
- ▶ A positive angle of rotation,  $\phi(x)$ , is a rotation of the cross section at  $x$  in a **right-hand-rule sense** about the  $x$  axis, as illustrated in Figure (4.2c).



**Figure (4.2): Torsional deformation; Sign convention for torsion**

## Assumptions

Following assumptions are made, while finding out shear stress in a circular shaft subjected to torsion:

- 1) The material of the shaft is uniform,
- 2) The twist along the shaft is uniform,
- 3) Normal cross – sections of the shaft, which were plane and circular before the twist, remain plane and circular even after the twist, and
- 4) All diameters of the normal cross – section, which were straight before the twist, remain straight with their magnitude unchanged, after the twist.

## Strain-Displacement Analysis

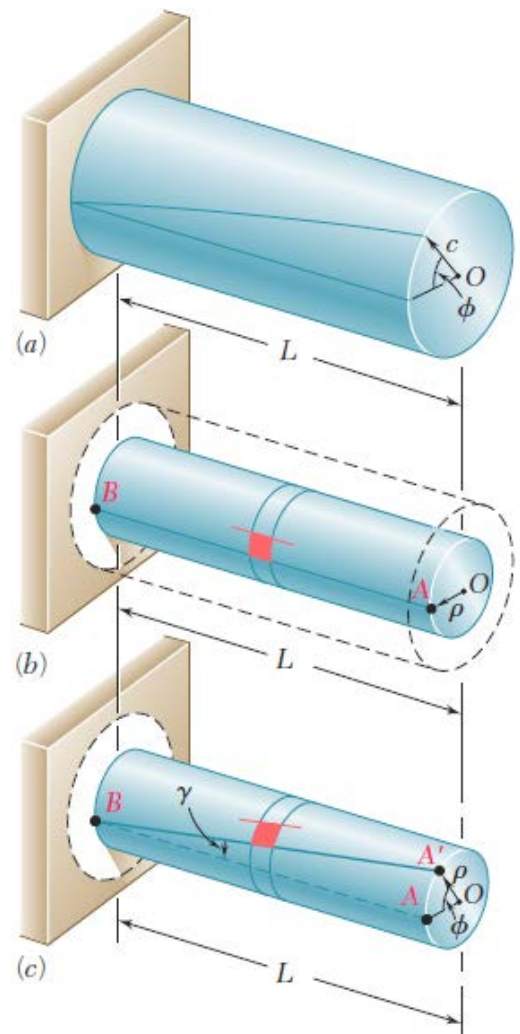
To determine the distribution of *shearing strains* in a circular shaft of length  $L$  and radius  $c$  that has been twisted through an angle  $\phi$  (Figure 4.3a).

- ▶ Detaching from the shaft a cylinder of radius  $r$ , we consider the small square element before any load is applied (Figure 4.3b).
- ▶ As the shaft is subjected to a torsional load, the element deforms into a rhombus (Figure 4.3c).
- ▶ The shearing strain  $\gamma$  in a given element is measured by the change in the angles formed by the sides of that element.
- ▶ Since the circles defining two of the sides of the element considered here remain unchanged, the shearing strain  $\gamma$  must be equal to the angle between lines AB and A'B.
- ▶ We observe from Figure (4.3c) that, for small values of  $\gamma$ , we can express the arc length AA' as:

$$AA' = \gamma L$$

- ▶ But, on the other hand, we have:

$$AA' = \rho \phi$$



- ▶ It follows that:

$$\rho\phi = \gamma L$$

Or

$$\gamma = \frac{\rho\phi}{L} \quad \dots (4.1)$$

Where  $\gamma$  and  $\phi$  are both expressed in radians.

- ▶ It follows from Eq. (4.1) that the shearing strain is maximum on the surface of the shaft, where  $\rho = c$ . We have:

$$\gamma_{max} = \frac{c \cdot \phi}{L} \quad \dots (4.2)$$

- ▶ Eliminating  $\phi$  from Eqs. (4.1) and (4.2), we can express the shearing strain  $\gamma$  at a distance  $r$  from the axis of the shaft as:

$$\gamma = \frac{\rho}{c} \gamma_{max} \quad \dots (4.3)$$

### STRESSES IN THE ELASTIC RANGE

- ▶ Let us now consider the case when the torque  $T$  is such that all shearing stresses in the shaft remain below the yield strength  $\tau_y$ .
- ▶ Recalling Hooke's law for shearing stress and strain:

$$\tau = G\gamma \quad \dots (4.4)$$

Where  $G$  is the modulus of rigidity or shear modulus of the material.

Multiplying both members of Eq. (4.3) by  $G$ , we write:

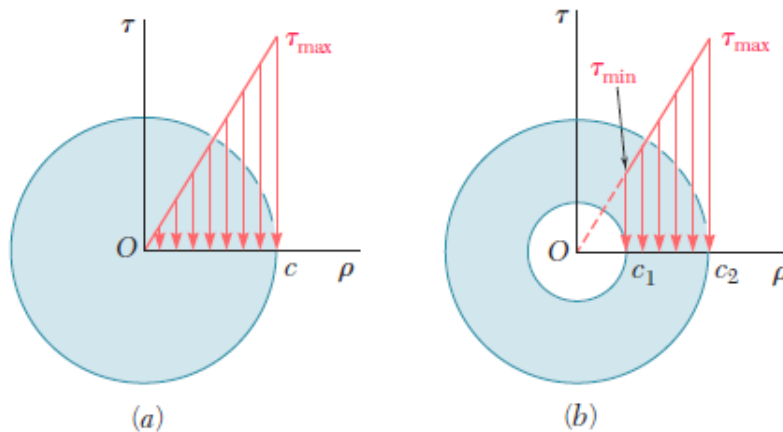
$$G\gamma = \frac{\rho}{c} G\gamma_{max} \quad \dots (4.5)$$

$$\tau = \frac{\rho}{c} \tau_{max} \quad \dots (4.6)$$

- ▶ Figure (4.4a) shows the stress distribution in a solid circular shaft of radius  $c$ , and Figure (4.4b) in a hollow circular shaft of inner radius  $c_1$  and outer radius  $c_2$ . From Eq. (4.6), we find that, in the latter case,

$$\tau_{min} = \frac{c_1}{c_2} \tau_{max} \quad \dots (4.7)$$

- The sum of the moments of the elementary forces exerted on any cross section of the shaft must be equal to the magnitude  $T$  of the torque exerted on the shaft:



**Figure (4.4): Distribution of shearing stresses.**

$$\int \rho(\tau dA) = T \quad \dots (4.8)$$

Substituting for  $\tau$  from (4.6) into (4.8), we write:

$$T = \int \rho(\tau dA) = \frac{\tau_{max}}{c} \int \rho^2 dA \quad \dots (4.9)$$

But,  $\int \rho^2 dA = J$  (*Polar moment of unertia:*), then:

$$T = \frac{\tau_{max} \cdot J}{c} \quad \dots (4.10)$$

or

$$\tau_{max} = \frac{T \cdot c}{J} \quad \dots (4.11)$$

- Substituting for  $\tau_{max}$  from Eqs.(4.11) into (4.6), we express the shearing stress at any distance  $\rho$  from the axis of the shaft as:

$$\tau = \frac{T\rho}{J} \quad \dots (4.12)$$

Equations (4.11) and (4.12) are known as *the elastic torsion formulas*.

- Substituting for ( $\gamma$ ) from Eq. (4.1) into Eq. (4.4) and then substituting for ( $\tau$ ) from Eq. (4.4) into Eq.(4.12), we express the angle of twist ( $\phi$ ) in term of the torque ( $T$ ):

$$\phi = \frac{Tl}{GJ} \quad \dots (4.13)$$

Recall from statics that the polar moment of inertia of a circle of radius ( $r$ ) is:

$$J = \frac{1}{2} \pi r^4$$

- ▶ In the case of a hollow circular shaft of inner radius ( $c_1$ ) and outer radius ( $c_2$ ), the polar moment of inertia is:

$$J = \frac{1}{2} \pi c_2^4 - \frac{1}{2} \pi c_1^4 = \frac{1}{2} \pi (c_2^4 - c_1^4)$$

- ▶ In Eq. 4.13, the torque- $T$ ,  $G$ , and  $J$  may each vary with  $x$  as shown in Figure (4.5). This derivative form of the torque-twist equation may be integrated over the length of the member to give:

$$\phi = \int_0^L \frac{T(x). dx}{G(x).J(x)} \quad \dots (4.14)$$

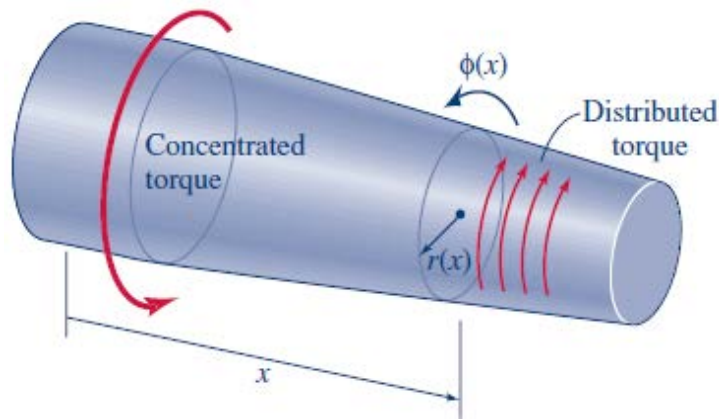


Figure (4.5)

### Examples

**Example (4.1):** A uniform shaft of radius  $r$  and length  $L$  is subjected to a uniform distributed external torque  $t_0$  (moment per unit length). (See Figure 4.6)

- Determine an expression for the maximum shear stress  $\tau_{max}$ .
- Determine an expression for the total twist angle  $\phi = \phi_L$ .

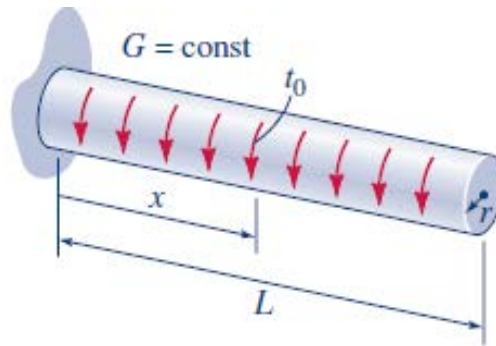


Figure (4.6)

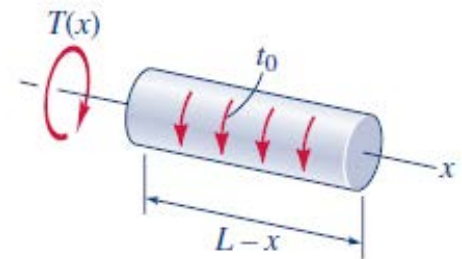
**Solution:**

a)

**Equilibrium:** On the section at  $x$ , the internal torque  $T(x)$  in the positive sense according to the right-hand rule is:

$$\sum M_x = 0$$

$$T(x) = t_o(L - x)$$



**Shear Stress:** The maximum shear stress occurs at  $x = 0$ , where  $T(0) = T_{max} = t_o(L)$ . Then, from the torsion formula, Eq. 4.11,

$$\tau_{max} = \frac{t_o(L) \cdot r}{\frac{\pi}{2} r^4} = \frac{2t_o(L)}{\pi r^3}$$

b)

**Torque-Twist:** From the torque-twist relationship, Eq. 4.14,

$$\phi_L = \int_0^L \frac{t_o(L - x) \cdot dx}{G \cdot J} = \frac{t_o}{G \cdot J} \int_0^L (L - x) \cdot dx$$

$$\phi_L = \frac{t_o}{G \cdot J} \left[ Lx - \frac{x^2}{2} \right]_0^L = \frac{t_o \cdot L^2}{2G \cdot J} = \frac{t_o \cdot L^2}{G \cdot \pi \cdot r^4}$$



**Example (4.2):** A hollow cylindrical steel shaft is **1.5 m** long and has inner and outer diameters respectively equal to **40** and **60 mm** (Figure 4.7).

- (a) What is the largest torque that can be applied to the shaft if the shearing stress is not to exceed **120 MPa**?
- (b) What is the corresponding minimum value of the shearing stress in the shaft?

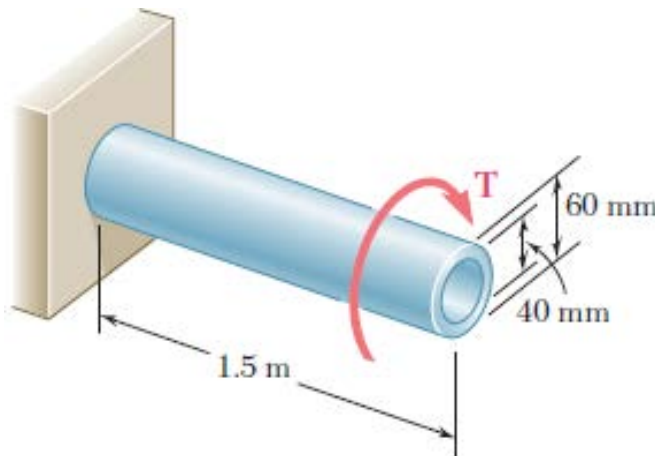


Figure (4.7)

**Solution:**

**(a) Largest Permissible Torque.**

The largest torque **T** that can be applied to the shaft is the torque for which  $\tau_{max} = 120$  MPa. Since this value is less than the yield strength for steel, we can use Eq. (4.11). Solving this equation for **T**, we have:

$$\tau_{max} = \frac{T \cdot c}{J} \rightarrow T = \frac{\tau_{max} \cdot J}{c}$$

$$J = \frac{1}{2} \pi c_2^4 - \frac{1}{2} \pi c_1^4 = \frac{1}{2} \pi (c_2^4 - c_1^4)$$

$$c_1 = \frac{40}{2} = 20 \text{ mm (or 0.02 m)}$$

$$c_2 = \frac{60}{2} = 30 \text{ mm (or 0.03 m)}$$

$$J = \frac{1}{2} \pi ((0.03)^4 - (0.02)^4) = 1.021 \times 10^{-6} \text{ m}^4$$

$$T = \frac{\tau_{max} \cdot J}{c} = \frac{(120 \times 10^6 \text{ Pa})(1.021 \times 10^{-6} \text{ m}^4)}{0.03} = 4.08 \text{ kN.m}$$

**(b) Minimum Shearing Stress.** The minimum value of the shearing stress occurs on the inner surface of the shaft. It is obtained from Eq. (4.7), which expresses that  $\tau_{min}$  and  $\tau_{max}$  are respectively proportional to  $c_1$  and  $c_2$ :

$$\tau_{min} = \frac{c_1}{c_2} \tau_{max}$$

$$\tau_{min} = \frac{c_1}{c_2} \tau_{max} = \frac{0.02}{0.03} (120) = 80 \text{ MPa}$$

**Example (4.3):** What is the minimum diameter of a solid steel shaft that will not twist through more than ( $3^\circ$ ) in a (6 m) length when subjected to a torque of (**14 kN.m**)? What maximum shearing stress is developed? Use **G = 83 GPa**.

**Solution:**

$$\phi = \frac{Tl}{GJ} \rightarrow 3 \left( \frac{\pi}{180} \right) = \frac{(14 \times 10^3)(6)}{(83 \times 10^9) \left( \frac{\pi}{2} r^4 \right)}$$

$$\rightarrow r = 59.23 \text{ mm or } d = 118.5 \text{ mm}$$

$$\tau_{max} = \frac{Tr}{J} = \frac{14 \times 10^6 (59.23)}{\frac{\pi}{2} (59.23)^4} = 42.9 \text{ MPa}$$

**Example (4.4):** Determine the length of the shortest (**2mm**) diameter bronze wire which can be twisted through two complete turns without exceeding a shearing stress of (**70 MPa**). Use **G = 83 GPa**.

**Solution:**

$$J = \frac{\pi}{32} d^4 = \frac{\pi}{32} (2)^4 = 1.571 \text{ mm}^4$$

$$\tau_{max} = \frac{T \cdot c}{J} \rightarrow T = \frac{\tau_{max} \cdot J}{c} \rightarrow$$

$$T = \frac{70 (\pi(1)^4)}{2(1)} \rightarrow T \cong 110 \text{ N.mm}$$

$$\phi = \frac{Tl}{GJ} \rightarrow 720 \left( \frac{\pi}{180} \right) = \frac{(110)(l)}{(83 \times 10^3)(1.571)}$$

$$\rightarrow l = 14895 \text{ mm or } 14.895 \text{ m}$$

**Example (4.5):** A compound shaft consisting of an aluminum segment and a steel segment is acted upon by two torque as shown in Figure (4.8). Determine the maximum permissible value of (T) subjected to the following conditions ( $\tau_s \leq 100 \text{ MPa}$ ,  $\tau_{al} \leq 70 \text{ MPa}$ , and the angle of rotation of the free end limited to  $(12^\circ)$ ). Use  $G_s = 83 \text{ GPa}$  and  $G_{al} = 28 \text{ GPa}$ .

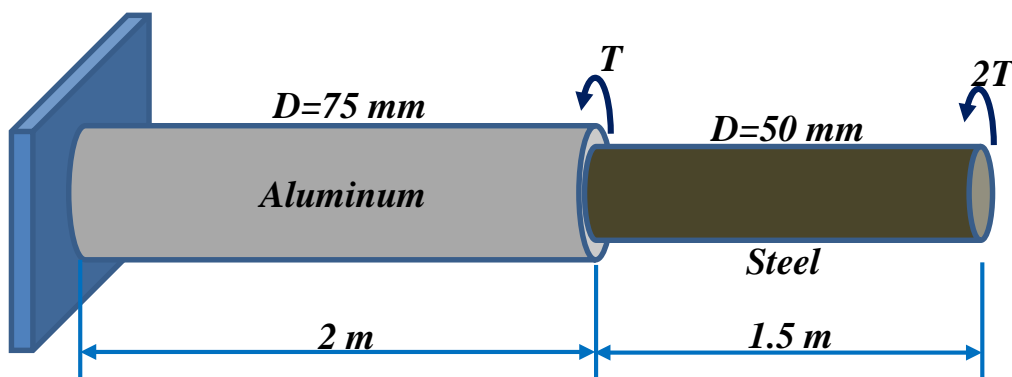


Figure (4.8)

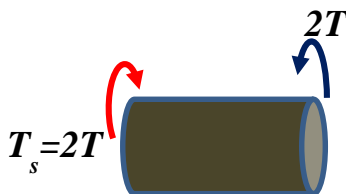
**Solution:**

$$\tau_s = \frac{T \cdot c}{J} \rightarrow T = \frac{\tau_s \cdot J}{c} \rightarrow T_s = \frac{\tau_s (\pi c^4)}{2 \cdot c} = \frac{\tau_s (\pi c^3)}{2} = \frac{100 (\pi (25)^3)}{2} \times 10^{-3}$$

$$= 2454.37 \text{ N.m}$$

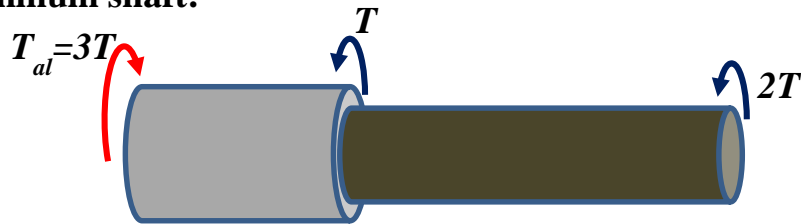
$$T_{al} = \frac{\tau_s (\pi c^3)}{2} = \frac{70 (\pi (37.5)^3)}{2} \times 10^{-3} = 5798.45 \text{ N.m}$$

**Section in steel shaft:**



$$T_s = 2T \rightarrow T = \frac{T_s}{2} = \frac{2454.37}{2} = 1227.2 \text{ N.m}$$

Section in aluminum shaft:



$$T_{al} = 3T \rightarrow T = \frac{T_{al}}{3} = \frac{5798.45}{3} = 1932.82 \text{ N.m}$$

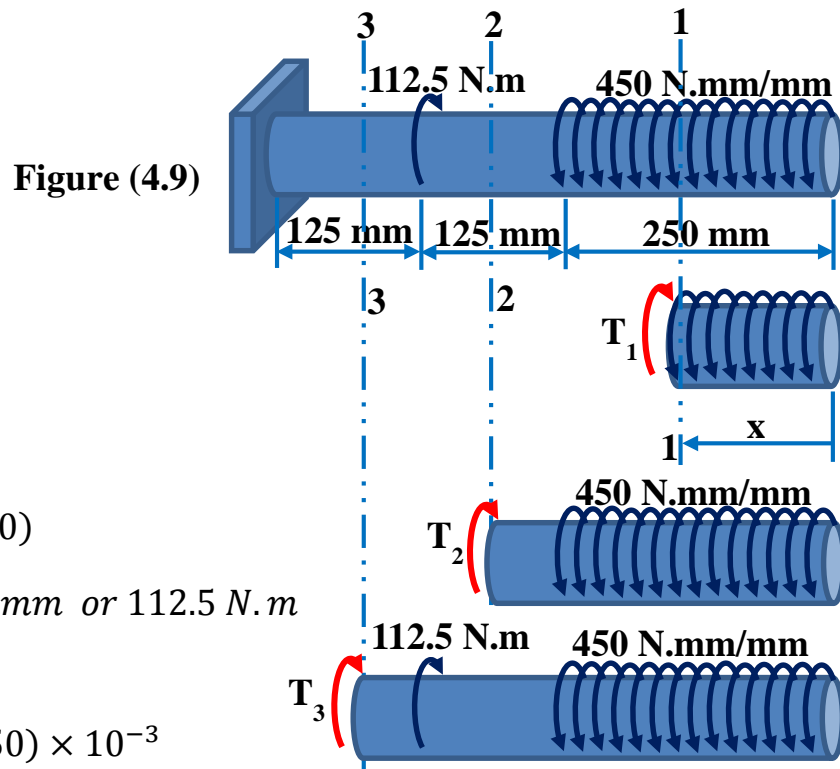
$$\phi_{free} = \phi_s + \phi_{al}$$

$$12 \left( \frac{\pi}{180} \right) = \frac{2T(1.5)}{(83 \times 10^9) \frac{\pi}{32} (50)^4 \times 10^{-12}} + \frac{3T(2)}{(83 \times 10^9) \frac{\pi}{32} (75)^4 \times 10^{-12}}$$

$$12 = 7.328 \times 10^{-3} T \rightarrow T = 1637.53 \text{ N.m}$$

Use  $T = 1227.2 \text{ N.m}$

**Example (4.6):** For the shaft shown in Figure (4.9), find the maximum shear stress and total angle of rotation. Assume  $G=83 \text{ GPa}$  and diameter of shaft (25 mm).



**Solution:**

Sec.(1-1) as F.B.D.

$$T = 450x$$

Sec.(2-2) as F.B.D.

$$T = 450(250)$$

$$T = 112500 \text{ N.mm or } 112.5 \text{ N.m}$$

Sec.(3-3) as F.B.D.

$$T + 112.5 = 450(250) \times 10^{-3}$$

$$\rightarrow T = 0$$

$$\therefore T = 112.5 \text{ N.m} \quad (\text{Control})$$

$$\therefore \tau_{max} = \frac{T \cdot c}{J} = \frac{(112.5 \times 10^3)(12.5)}{\frac{\pi}{2}(12.5)^4} = 36.67 \text{ MPa}$$

$$\phi_{total} = \sum \phi_i = 0 + \phi_1 + \phi_2$$

$$\phi_1 = \int_0^{250} \frac{(450x)dx}{GJ} = \frac{450x^2}{2GJ} \Big|_0^{250} = \frac{450(250)^2}{2(84 \times 10^3) \left(\frac{\pi}{32}(25)^4\right)}$$

$$\phi_1 = 4.365 \times 10^{-3} \text{ rad.}$$

$$\phi_2 = \frac{112.5 \times 10^3(125)}{(84 \times 10^3) \left(\frac{\pi}{32}(25)^4\right)} = 4.365 \times 10^{-3} \text{ rad.}$$

$$\phi_{total} = [4.365 \times 10^{-3} + 4.365 \times 10^{-3}] \frac{180}{\pi} = 0.5^\circ$$

**Example (4.7):** Find the total angle of rotation for the prismatic shaft shown in Figure (4.10), which is subjected to distributed torsional moment ( $T_{(x)}=kx \text{ N.mm/mm}$ ),  $GJ$  and  $k$  are constants.

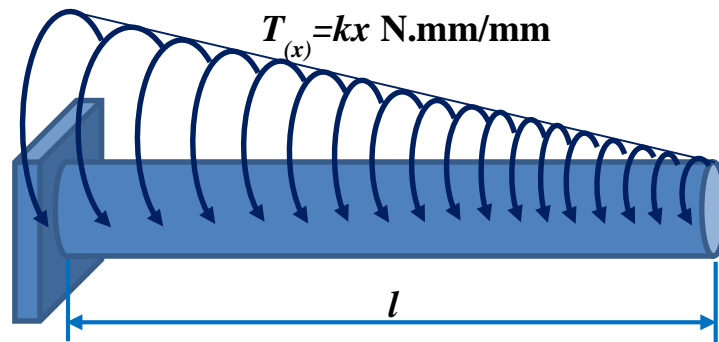
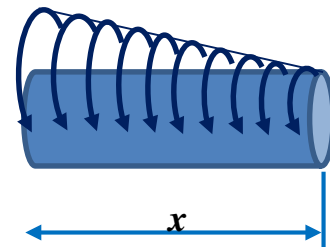


Figure (4.10)

**Solution:**

$$T = \int_0^x (kx)dx = \frac{kx^2}{2}$$



$$\phi = \int_0^l \frac{T}{GJ} dx = \frac{1}{GJ} \int_0^l \frac{kx^2}{2} dx$$

$$\therefore \phi = \frac{1}{2GJ} \left[ \frac{kx^3}{3} \right]_0^l = \frac{kl^3}{6GJ}$$

**To find max. Shear stress:**

$$\tau_{max} \frac{16T}{\pi d^3} = \frac{16kx^2}{2\pi d^3} \Bigg|_{x=l} = \frac{16kl^2}{2\pi d^3} = \frac{8kl^2}{\pi d^3}$$

**Example (4.8):** For the non – prismatic shaft shown in Figure (4.11). Find the total rotation.

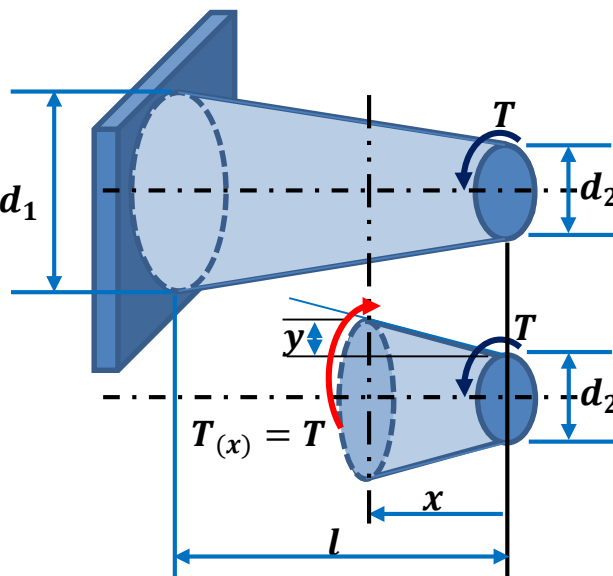
**Solution:**

$$\frac{\frac{d_1 - d_2}{2}}{l} = \frac{y}{x}$$

$$\rightarrow y = \frac{d_1 - d_2}{2l} x$$

$$\phi = \int_0^l \frac{T(x) dx}{GJ(x)}$$

Figure (4.11)



$$J(x) = \frac{\pi}{32} (D(x))^4$$

$$D(x) = 2y + d_2 = \frac{d_1 - d_2}{l} x + d_2$$

$$J(x) = \frac{\pi}{32} \left[ \frac{d_1 - d_2}{l} x + d_2 \right]^4$$

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$$\phi = \int_0^l \frac{T dx}{G \frac{\pi}{32} \left[ \frac{d_1 - d_2}{l} x + d_2 \right]^4} = \frac{32T}{G\pi} \int_0^l \left[ \frac{d_1 - d_2}{l} x + d_2 \right]^{-4} dx$$

Multiply by  $\frac{\frac{d_1 - d_2}{l}}{\frac{d_1 - d_2}{l}}$ , get:

$$\begin{aligned} \phi &= \frac{32T}{G\pi(d_1 - d_2)} \left[ \left[ \frac{d_1 - d_2}{l} x + d_2 \right]^{-3} \cdot \frac{1}{-3} \right]_0^l \\ &= \frac{32T}{-3G\pi(d_1 - d_2)} \left[ (d_1)^{-3} - (d_2)^{-3} \right] \\ \phi &= \frac{32T}{3G\pi} \left[ \frac{(d_1)^2 + d_1 d_2 + (d_2)^2}{(d_1)^3 (d_2)^3} \right] \end{aligned}$$