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Al-Mustaqbal University College Air Conditioning and Refrigeration Techniques Engineering Department

Strength of Materials

Second Stage

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Murtadha Al-Masoudy HOOKE'S LAW FOR PLANE STRESS

Poisson's Ratio

□ <u>THE RELATIONSHIP BETWEEN E AND G</u>:

For a *uniaxial stress state*, σ_x , (Figure 6.7) the linear relationship between stress and strain, ϵ_x , was given by Hooke's Law:

$$\epsilon_x = \frac{\sigma_x}{E}$$

Poisson's ratio, v, relates the transverse strain, ϵ_v , to ϵ_x :

 $\epsilon_y = \epsilon_z = -v\epsilon_x = -v\frac{\sigma_x}{E}$

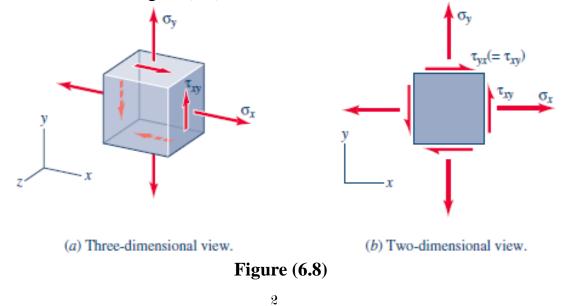
Figure 6.7

Original specimen

For linearly elastic materials, the shear modulus G and Young's modulus E are related by the equation:

$$G = \frac{E}{2(1+\nu)}$$

- ► For this equation to apply, the material must not only be linearly elastic, it must also be **isotropic**, and that is, *its material properties like E and v must be independent of orientation in the body*.
- □ Plane Stress. A body that is subjected to a two-dimensional state of stress with $\sigma_z = \tau_{xz} = \tau_{yz} = 0$, is said to be in a state of plane stress. An element in plane stress is shown in Figure (6.8).



If the material of which the body is composed is linearly elastic and isotropic, the effects of stresses σ_x , σ_y , and τ_{xy} can be superposed, giving *Hooke's Law for plane stress:*

$$\epsilon_x = \frac{1}{E} (\sigma_x - \nu \sigma_y)$$
$$\epsilon_y = \frac{1}{E} (\sigma_y - \nu \sigma_x)$$
$$\gamma_{xy} = \frac{1}{E} \tau_{xy}$$

Generalized Hooke's Law for Isotropic Materials

Let the body be subjected to stresses σ_x , σ_y , σ_z , τ_{xy} , τ_{xz} and τ_{yz} as shown in Figure (6.9).

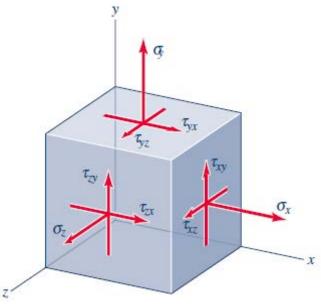


Figure (6.9)

Figure (6.10) illustrate the strains produced separately by the three normal stresses, σ_x , σ_y , and σ_z .

By the superposition principle, the total extensional strains are given by:

$$\epsilon_x = \frac{1}{E} \left[\sigma_x - v \left(\sigma_y + \sigma_z \right) \right]$$
$$\epsilon_y = \frac{1}{E} \left[\sigma_y - v \left(\sigma_x + \sigma_z \right) \right]$$

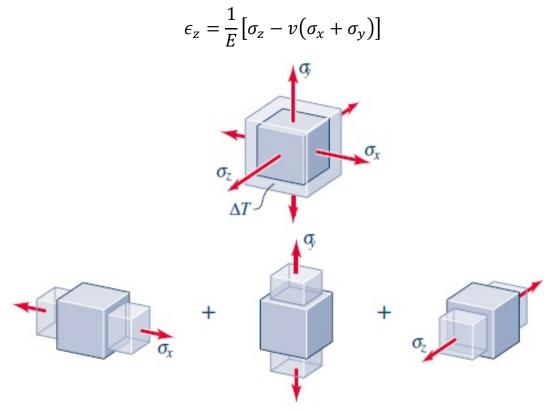


Figure (6.10)

► For an isotropic linearly elastic material, the shear stresses are related to the shear strains by the following equations:

$$\gamma_{xy} = \frac{1}{G} \tau_{xy}, \ \gamma_{xz} = \frac{1}{G} \tau_{xz}, \ \gamma_{yz} = \frac{1}{G} \tau_{yz}$$

These shear strains are illustrated in Figure (6.11).

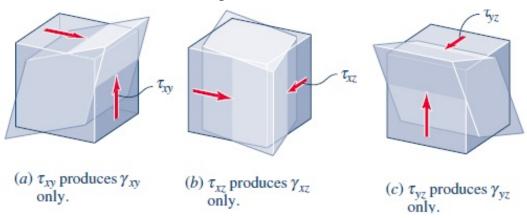


Figure (6.11): Illustration of shear strains.

Solving the above equations for the stresses in terms of the strains, we get:

$$\sigma_x = \frac{E}{(1+v)(1-2v)} [(1-v)\epsilon_x + v(\epsilon_y + \epsilon_z)]$$

$$\sigma_y = \frac{E}{(1+v)(1-2v)} [(1-v)\epsilon_y + v(\epsilon_x + \epsilon_z)]$$

$$\sigma_z = \frac{E}{(1+v)(1-2v)} [(1-v)\epsilon_z + v(\epsilon_y + \epsilon_x)]$$

and

$$au_{xy}=G\gamma_{xy}$$
 , $au_{xz}=G\gamma_{xz}$, $au_{zy}=G\gamma_{zy}$

Examples

Example (6.5): A prismatic bar of circular cross – section is loaded by tensile force (P=85 kN). The bar has length (3 m) and diameter (30 mm). It is made of aluminum with modulus of elasticity (70 GPa) and Poisson's ratio (1/3). Calculate the elongation, the decrease in diameter and the change in volume.

Solution:

$$\delta = \varepsilon_x \cdot l \quad and \quad \varepsilon_x = \frac{\sigma_x}{E}$$

$$\sigma_x = \frac{P}{A} = \frac{85 \times 10^3}{\frac{\pi}{4} (30)^2} = 120.25 \, MPa$$

$$\varepsilon_x = \frac{\sigma_x}{E} = \frac{120.25}{70 \times 10^3} = 0.00172$$

$$\delta = \varepsilon_x \cdot l = 0.00172(3 \times 10^3) = 5.16 \, mm$$

$$\Delta d = \varepsilon_y \cdot d$$

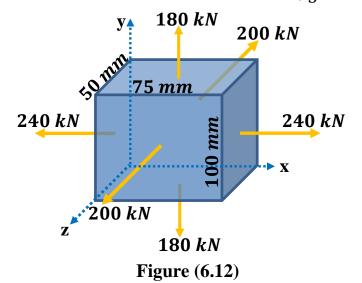
$$\varepsilon_y = -v \frac{\sigma_x}{E} = -\frac{1}{3} \frac{(120.25)}{(70 \times 10^3)} = -0.00057$$

$$\Delta d = -0.00057(30) = -0.0171 \, mm$$

Change in volume (ΔV)=final volume – initial volume

$$\Delta V = \left[(30 - 0.0171)^2 \left(\frac{\pi}{4}\right) (3000 + 5.16) \right] - \left[(30)^2 \left(\frac{\pi}{4}\right) (3000) \right] = 1226.5 \ mm^3$$

Example (6.6): Find a single force in x –direction that gives the same change in the direction parallel to x, for shown Figure (6.12). Take $v = \frac{1}{3}$ and E = 70 GPa.



Solution:

$$\varepsilon_{x} = \frac{\sigma_{x}}{E} - v \frac{\sigma_{y}}{E} - v \frac{\sigma_{z}}{E}$$

$$\sigma_{x} = \frac{240 \times 10^{3}}{50(100)} = 48 MPa$$

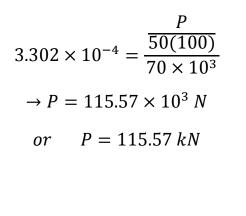
$$\sigma_{y} = \frac{180 \times 10^{3}}{50(75)} = 48 MPa$$

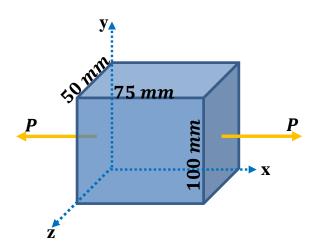
$$\sigma_{z} = \frac{200 \times 10^{3}}{75(100)} = 26.67 MPa$$

$$\varepsilon_{x} = \frac{48}{70 \times 10^{3}} - \frac{1}{3} \frac{48}{(70 \times 10^{3})} - \frac{1}{3} \frac{26.67}{(70 \times 10^{3})} = 3.302 \times 10^{-4}$$

Uniaxial stress state is:

$$\varepsilon_{\chi} = \frac{\sigma_{\chi}}{E}$$





Example (6.7): A uniform bar of length (l), cross – sectional area (A), and unit mass (ρ) is suspended vertically from one end as shown in Figure (6.13). Show that its total elongation is ($\delta = \frac{\rho g l^2}{2E}$). If the total mass of the bar is (M), show also that ($\delta = \frac{Mg l}{2EA}$).

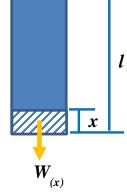


Figure (6.13)

Solution:

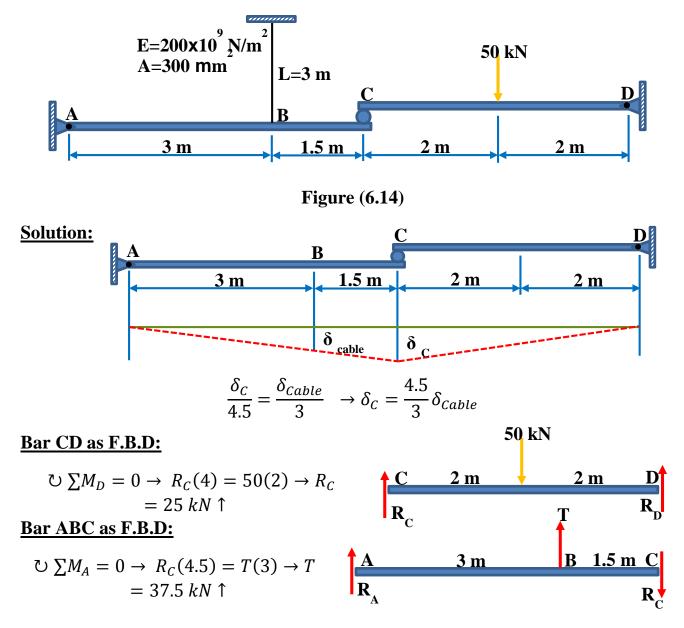
$$\rho_{(x)} = W_{(x)} = \gamma . Vol. = \rho(g)(A)(x)$$
$$\delta = \int_0^l \frac{\rho_{(x)}}{EA} dx = \int_0^l \frac{\rho(g)(A)}{EA} dx = \frac{\rho(g)}{E} \left[\frac{x^2}{2}\right]_0^l = \frac{\rho g l^2}{2E}$$

Total Mass=M, let total weight=M.g.l

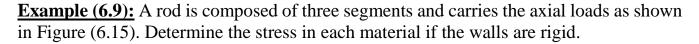
unit weight
$$W_{(x)} = \frac{M \cdot g}{l}(x)$$

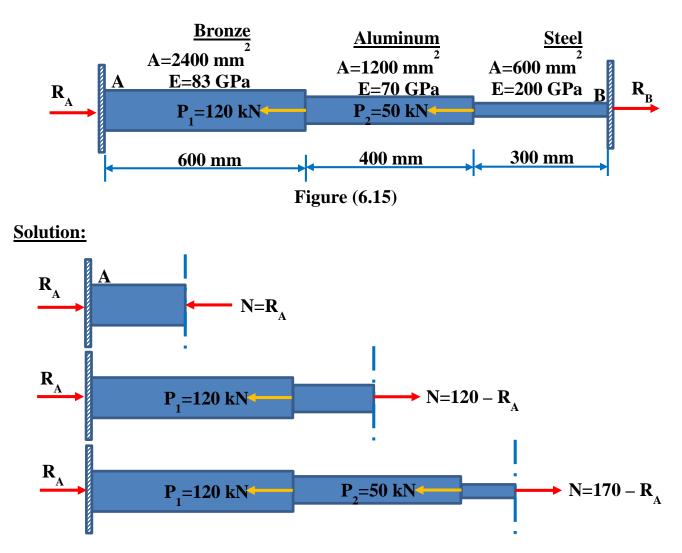
$$\delta = \int_0^l \frac{M \cdot g}{EA}(x) dx = \int_0^l \frac{M \cdot g}{EA(l)}(x) dx$$
$$\delta = \frac{M \cdot g}{EA(l)} \left[\frac{x^2}{2}\right]_0^l = \frac{M \cdot g(l)}{2EA}$$

Example (6.8): The rigid bars shown in Figure (6.14) are separated by a roller at point (C) and pinned at point (A) and (D). A steel rod at point (B) helps support the load of (50 kN). Compute the vertical displacement of the roller at point (C).



$$\delta_{Cable} \frac{Pl}{EA} = \frac{37.5 \times 10^3}{200 \times 10^3 (300)} = 1.875 \ mm$$
$$\delta_C = \frac{4.5}{3} \delta_{Cable} = \frac{4.5}{3} (1.875) = 2.813 \ mm$$





From equilibrium:

$$R_A + R_B = P_1 + P_2 \rightarrow R_A + R_B = 170 \dots \dots \dots \dots \dots (1)$$

From compatibility:

$$\delta_{\text{Br.}} + \delta_{\text{Al.}} + \delta_{\text{St.}} = 0$$

$$\frac{R_A \times 10^3(600)}{83 \times 10^3(2400)} + \frac{(120 - R_A) \times 10^3(400)}{70 \times 10^3(1200)} + \frac{(170 - R_A) \times 10^3(300)}{83 \times 10^3(600)} = 0$$
$$\rightarrow R_A = 96.99 \cong 97 \ kN$$

Sub. in Equ.(1),get:

$$R_A + R_B = 170 \rightarrow R_B = 170 - 97 = 73 \ kN$$

$$\sigma_{Br.} = \frac{P}{A} = \frac{97 \times 10^3}{2400} = 40.42 \ MPa$$

$$\sigma_{Al.} = \frac{P}{A} = \frac{(120 - 97) \times 10^3}{1200} = 19.17 \ MPa$$

$$\sigma_{St.} = \frac{P}{A} = \frac{73 \times 10^3}{600} = 121.67 \ MPa$$

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