# **Applications of Thermodynamics to Flow Processes**

Flow processes inevitably result from pressure gradients within the fluid. Moreover, temperature, velocity, and even concentration gradients may exist within the flowing fluid. This contrasts with the uniform conditions that prevail at equilibrium in closed systems. The distribution of conditions in flow systems requires that properties be attributed to point masses of fluid. Thus we assume that intensive properties, such as density, specific enthalpy, specific entropy, etc., at a point are determined solely by the temperature, pressure, and composition at the point, uninfluenced by gradients that may exist at the point. Moreover, we assume that the fluid exhibits the same set of intensive properties at the point as though it existed at equilibrium at the same temperature, pressure, and composition.

#### 1. TURBINES (EXPANDERS)

The expansion of a gas in a nozzle to produce a high-velocity stream is a process that converts internal energy into kinetic energy. This kinetic energy is in turn converted into shaft work when the stream impinges on blades attached to a rotating shaft. Thus a turbine (or expander) consists of alternate sets of nozzles and rotating blades through which vapor or gas flows in a steady-state expansion process whose overall effect is the efficient conversion of the internal energy of a high-pressure stream into shaft work. When steam provides the motive force as in a power plant, the device is called a turbine; when a high-pressure gas, such as ammonia or ethylene in a chemical or petrochemical plant, is the working fluid, the device is often called an expander. The process for either case is shown in Fig. below



Figure 7.3 Steady-state flow through a turbine or expander

Equations (2.31) and (2.32) are appropriate energy relations. However, the potential energy term can be omitted, because there is little change in elevation. Moreover, in any properly designed turbine, heat transfer is negligible and the inlet and exit pipes are sized to make fluid velocities roughly equal. Equations (2.31) and (2.32) therefore reduce to:

$$\Delta \left(H + \frac{1}{2}u^2 + zg\right)\dot{m} = \dot{Q} + \dot{W}_s \tag{2.31}$$

$$\Delta H + \frac{\Delta u^2}{2} + g \,\Delta z = Q + W_s \tag{2.32a}$$

$$\dot{W}_s = \dot{m} \Delta H = \dot{m}(H_2 - H_1)$$
 (7.13)

$$W_s = \Delta H = H_2 - H_1 \tag{7.14}$$

alone does not allow any calculations to be made. However, if the fluid in the turbine undergoes an expansion process that is reversible as well as adiabatic, then the process is isentropic, and  $S_2 = S_1$ . This second equation allows

determination of the final state of the fluid and hence of  $H_2$ . For this special case, W, is given by Eq. (7.14), written:

$$W_s(\text{isentropic}) = (\Delta H)_s$$
 (7.15)

The shaft work |Ws|, (*isentropic*)| is the maximum that can be obtained from an adiabatic turbine with given inlet conditions and given discharge pressure. Actual turbines produce less work, because the actual expansion process is irreversible. We therefore define a *turbine eficiency* as:

$$\eta \equiv \frac{W_s}{W_s(\text{isentropic})}$$

where  $W_s$ , is the actual shaft work. By Eqs. (7.14) and (7.15),

1.122

$$\eta = \frac{\Delta H}{(\Delta H)_S} \tag{7.16}$$

Values of  $\eta$  for properly designed turbines or expanders usually range from 0.7 to 0.8. Figure 7.4 shows an H<sub>s</sub> diagram on which are compared an actual expansion process in a turbine and the reversible process for the same intake conditions and the same discharge pressure. The reversible path is a vertical line of constant entropy from point 1 at the intake pressure P<sub>1</sub> to point 2' at the discharge pressure P<sub>2</sub>. The line representing the actual irreversible process starts also from point 1, but is directed downward and to the right, in the direction of increasing entropy. Since the process is adiabatic, irreversibilities cause an increase in entropy of the fluid. The process terminates at point 2 on the isobar for P<sub>2</sub>. The more irreversible the process, the further this point lies to the right on the P<sub>2</sub> isobar, and the lower the efficiency  $\eta$  of the process.



Figure 7.4 Adiabatic expansion process in a turbine or expander

## Two-Phase Liquid / Vapor Systems

When a system consists of saturated-liquid and saturated-vapor phases coexisting in equilibrium, the total value of any extensive property of the two-phase system is the sum of the total properties of the phases. Written for the volume, this relation is:

$$nV = n^l V^l + n^v V^v$$

where V is the system volume on a molar basis and the total number of moles is  $n = n^{l} + n^{v}$ . Division by *n* gives:

$$V = x^l V^l + x^v V^v$$

where  $x^{l}$  and  $x^{u}$  represent the fractions of the total system that are liquid and vapor. Since  $x^{l} = 1 - x^{v}$ ,

$$V = (1 - x^{\nu})V^l + x^{\nu}V^{\nu}$$

In this equation the properties V,  $V^l$ , and  $V^v$  may be either molar or unit-mass values. The mass or molar fraction of the system that is vapor  $x^v$  is called the *quality*. Analogous equations can be written for the other extensive thermodynamic properties. All of these relations are represented by the generic equation:

$$M = (1 - x^{v})M^{l} + x^{v}M^{v}$$
(6.73a)

where M represents V, U, H, S, etc. An alternative form is sometimes useful:

$$M = M^l + x^v \Delta M^{lv} \tag{6.73b}$$

# Example 7.6

A steam turbine with rated capacity of 56,400 kW (56,400 kJ·s<sup>-1</sup>) operates with steam at inlet conditions of 8600 kPa and 500°C, and discharges into a condenser at a pressure of 10 kPa. Assuming a turbine efficiency of 0.75, determine the state of the steam at discharge and the mass rate of flow of the steam.

### Solution 7.6

At the inlet conditions of 8600 kPa and 500°C, the steam tables provide:

$$H_1 = 3391.6 \text{ kJ} \cdot \text{kg}^{-1}$$
  $S_1 = 6.6858 \text{ kJ} \cdot \text{kg}^{-1} \cdot \text{K}^{-1}$ 

If the expansion to 10 kPa is isentropic, then,  $S'_2 = S_1 = 6.6858 \text{ kJ} \cdot \text{kg}^{-1} \cdot \text{K}^{-1}$ . Steam with this entropy at 10 kPa is wet. Applying the "lever rule" [Eq. (6.96b), with M = S and  $x^{\nu} = x'_2$ ], the quality is obtained as follows:

$$S'_2 = S^l_2 + x'_2(S^v_2 - S^l_2)$$

Then,  $6.6858 = 0.6493 + x'_2(8.1511 - 0.6493)$   $x'_2 = 0.8047$ 

This is the quality (fraction vapor) of the discharge stream at point 2'. The enthalpy  $H'_2$  is also given by Eq. (6.96b), written:

$$H_2' = H_2^l + x_2'(H_2^v - H_2^l)$$

Thus,

 $H'_2 = 191.8 + (0.8047)(2584.8 - 191.8) = 2117.4 \text{ kJ} \cdot \text{kg}^{-1}$  $(\Delta H)_S = H'_2 - H_1 = 2117.4 - 3391.6 = -1274.2 \text{ kJ} \cdot \text{kg}^{-1}$ 

and by Eq. (7.16),

$$\Delta H = \eta \, (\Delta H)_S = (0.75)(-1274.2) = -955.6 \, \text{kJ} \cdot \text{kg}^{-1}$$

Whence,  $H_2 = H_1 + \Delta H = 3391.6 - 955.6 = 2436.0 \text{ kJ} \cdot \text{kg}^{-1}$ 

Thus the steam in its actual final state is also wet, with its quality given by:

$$2436.0 = 191.8 + x_2(2584.8 - 191.8) \qquad x_2 = 0.9378$$

Then  $S_2 = 0.6493 + (0.9378)(8.1511 - 0.6493) = 7.6846 \text{ kJ} \cdot \text{kg}^{-1} \cdot \text{K}^{-1}$ 

This value may be compared with the initial value of  $S_1 = 6.6858$ .

The steam rate  $\dot{m}$  is given by Eq. (7.13). For a work rate of 56,400 kJ·s<sup>-1</sup>,

$$\dot{W}_s = -56,400 = \dot{m}(2436.0 - 3391.6)$$
  $\dot{m} = 59.02 \text{ kg} \cdot \text{s}^{-1}$ 

Table F.2 Superheated Steam, SI Units (Continued)

EMPERATURE: T kelvins (TEMPERATURE: t°C)

923.15 (650) 50.313 3355.7 7.139 7.139 7.139 7.139 7.139 7.139 7.139 7.139 7.1155 7.1155 7.1155 7.1155 7.1035 7.1035 7.1035 7.1035 7.1035 7.1035 7.1035 7.1035 7.0919 7.0919 7.0919 7.0919 7.0806 7.0919 7.0695 7.06 898.15 (025) (025) 48.747 48.747 598.1 7.0739 47.544 46.397 7.0614 46.397 7.0492 46.397 7.0492 46.397 7.0492 44.255 693.1 7.0373 44.255 693.3 7.0029 44.255 688.4 7.0029 44.255 688.4 7.0029 7.0029 688.4 688.4 7.0029 7.0029 7.0029 7.0029 7.0029 688.4 688.4 688.4 7.0029 873.15 (600) (600) 251.1 637.9 537.9 537.9 537.9 6.9936 6.9936 6.9936 6.9813 6.9813 6.9813 6.9813 6.9813 6.9813 6.9828 6.9828 6.9828 6.9824 6.9624 6.9574 6.9574 6.9574 6.9573 6.92436 6.92436 6.92436 6.92436 6.92436 6.92436 6.92436 6.92436 6 823.15 (550) (550) (550) (550) (57.8 6.8646 42.839 517.8 515.8 6.8516 6.8516 6.8516 6.8516 6.8330 40.782 515.8 6.8330 6.8330 6.8330 6.8330 6.8265 6.8265 6.8265 6.8265 6.8265 6.8205 6.8205 6.7300 6.7790 6.7790 6.7790 798.15 (525) (525) (525) (525) 8.7900 8.7900 8.7769 8.7760 8.7700 8.77600 8.77760 8.7760 8.7760 8.7760 8.7760 8.77 748.15 (475) (475) (475) 334.5 6.6311 37.887 334.5 6.6311 334.6 6.6037 36.011 331.9 6.6037 36.011 329.2 6.6037 36.011 329.2 6.5004 32.136 6.5504 32.136 6.5504 32.136 6.5517 33.495 833.405 83 sat. vap. 757.0 5.7338 5.7338 5.7203 1.391 315.2 326.6 326.6 325.7 325.7 336.1 335.1 336.1 345.4 345.4 345.4 345.4 345.4 345.4 345.4 345.4 345.4 345.4 345.4 345.4 357.1 354.6 355.7 354.6 355.7 354.6 355.7 354.6 355.7 356.7 liq. Sat *Р*/кРа т<sup>ыж</sup>/К (t<sup>ык</sup>/°C) 669.85(296.70) 671.54(298.39) 573.21(300.06) 574.85(301.70) 576.46(303.31) 578.04(304.89) 578.04(306.44) 9600 581.12(307.97)